

# Electric Field Measurements

- Types of electric field measurements
- Double Probe Measurements
  - Theory
  - Measurements in the ionosphere
  - Measurements in the magnetosphere
  - The THEMIS electric field experiment
- Further Reading:
  - Bonnell et al., The THEMIS EFI instrument, *Space Science Reviews*, 2008.
  - Maynard, N. C., 1998 Electric field measurements in moderate to high density space plasmas with passive double probes, in *Measurement Techniques in Space Plasmas*, AGU Geophys. Monogr. 103, p., 13, 1998.
  - Paschmann, G., et al., The electron drift technique for measuring electric and magnetic fields, in *Space Plasmas*, AGU Geophys. Monogr. 103, p., 39, 1998.
  - Pedersen et al., Electric field measurements in a tenuous plasma with spherical double probes, in *Measurement Techniques in Space Plasmas*, AGU Geophys. Monogr. 103, p. 1, 1998.
  - Harvey, P. R., et al., The electric field instrument on the POLAR satellite, *Space Sci. Rev.*, 71, 583, 1995.
  - Gustafsson, G., et al., The electric field and wave experiment for the Cluster mission, *Space Sci. Rev.*, 79, 137, 2008.
  - Whipple, E. C., Potentials of surfaces in space, *Rep. Prog. Phys.*, 44, 1197, 1981.
  - Laakso H., and A. Pedersen, Satellite photoemission characteristics, in *Materials in a space environment*, edited by H. T. D. Guyenne, pp 361-365, ESA SP-368, ESTEC, Noordwijk, 1994.
  - Grard, R. J. L., Properties of the satellite photoelectron sheath derived from photoemission laboratory measurements, *J. Geophys. Res.*, 78, 2885, 1973.

# Electric Fields, Overview

Type of E-field Instr.	Basic Principle	Output quantity / limits
<b>Double Probe</b>	direct potential measurement	wave form up to 10Hz <i>0.5mV/m (10Hz ... DC)</i> compressed data up to 100kHz <i>50nV/m/Hz</i>
<b>Electron Drift Instrument</b>	<b>ExB</b> drift of emitted electrons (displacement or time of flight measurement)	E field normal to B <i>0.05mV/m (10Hz ... DC)</i>
All particle Instruments	E-field derived from measured velocity distribution and B-field	E field normal to B <i>0.1..1mV/m depends on plasma (spin period... DC)</i>

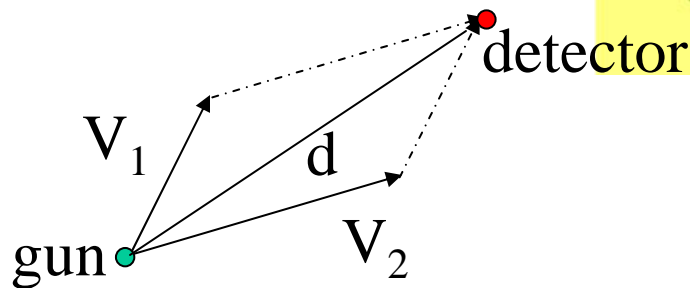
# Electron Drift Instrument

Measurement of displacement

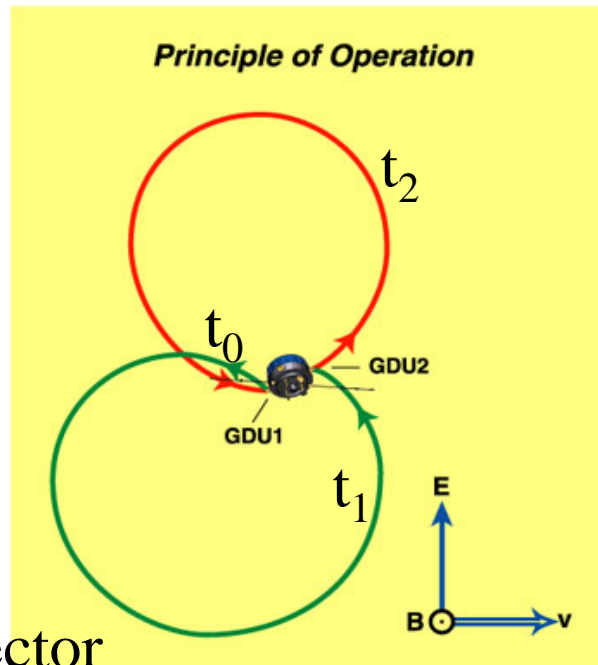
$$\mathbf{v}_D = \frac{\mathbf{E} \times \mathbf{B}}{B^2}$$

$$\mathbf{d} = \mathbf{v}_D T_g = \frac{\mathbf{E} \times \mathbf{B}}{B^2} T_g$$

$$d[m] = 3.57 \cdot 10^4 \frac{E_{\perp}[mV/m]}{B^2[nT]}$$



E-field measurement - EDI (1)  
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Measurement of time of flight

$$t_1 = t_0 + T_g(1 + \mathbf{v}_D/\mathbf{v}_e)$$

$$t_2 = t_0 + T_g(1 - \mathbf{v}_D/\mathbf{v}_e)$$

$$\mathbf{v}_D = \frac{\Delta t}{2T_g} \mathbf{v}_e$$

$$|\mathbf{B}| = \frac{2\pi m}{eT_g}$$

$$T_g = ((t_1 - t_0) + (t_2 - t_0))/2$$

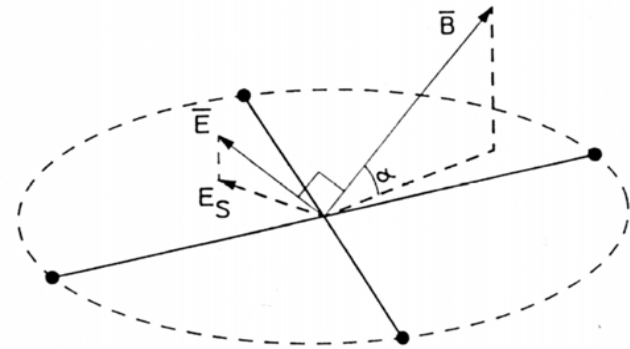
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# *Electron Drift Instrument*

- Principle of Operation
  - E-field normal to  $\mathbf{B}$  derived by measurement of  $\mathbf{E} \times \mathbf{B}$  drift of emitted electrons
  - Two modes: (1) measurement of displacement for high fields, (2) time of flight measurement for moderate fields.
  - In mode (2) magnitude of  $B$  can be derived by gyration period
  - Gradient  $\mathbf{B}$  drift can be separated by different energies of test electrons
- Parameter
  - Mode and Applicability depends on  $\mathbf{B}$ -field magnitude and variability
  - Accuracy: 0.05 mV/m (absolute)
- Application
  - Ground application: not applicable
  - Space application: GEOS, Geotail, proof of principle  
Equator-S, Cluster fully operational

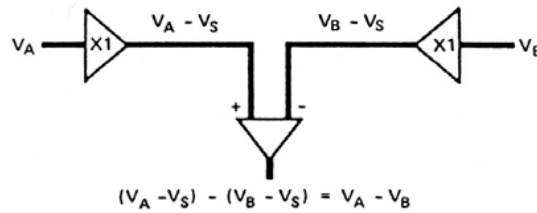
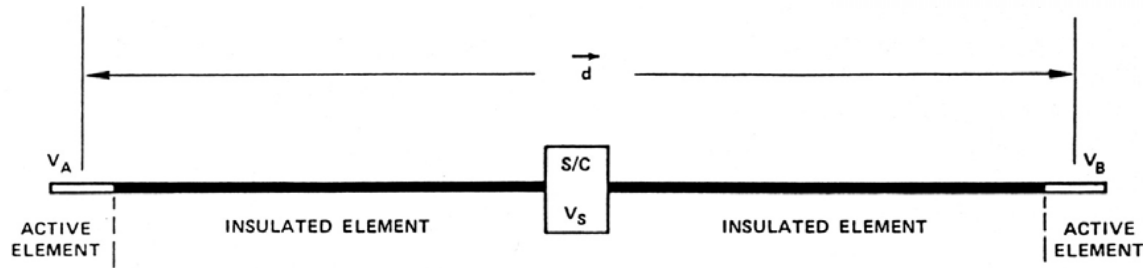
# Double Probe

$$(\Phi_1 - \Phi_2) / |d| = (\mathbf{E} + \mathbf{v} \times \mathbf{B}) \cdot \mathbf{d}$$



b)

VIKING (100m)  
CLUSTER (100m)  
POLAR (100m+130m)



[Pedersen et al., 1998]

[N.C.Maynard, 1998]

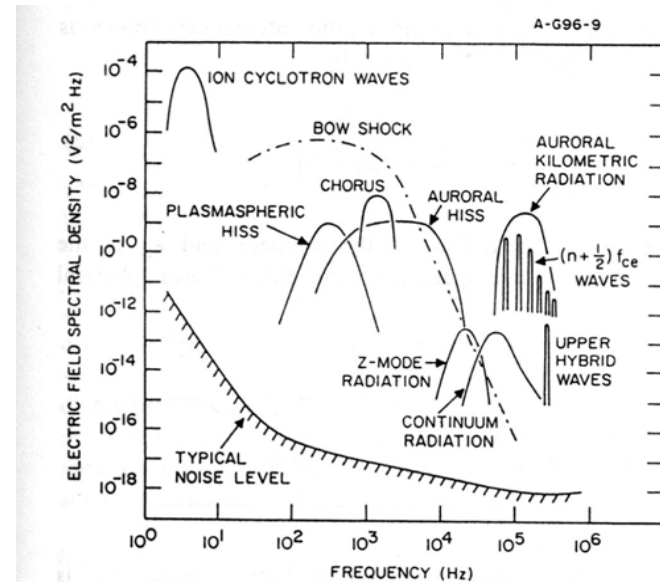
E-field measurement - Double Probe (1)

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# Double probe

- Principle of Operation
  - Direct measurement
  - Only method which can be used to measure  $E_{\parallel}$
- Parameter
  - Noise at 1Hz:  $10^{-5}$  V/m/sqrt(Hz)
  - Noise at 100 kHz:  $10^{-9}$  V/m/sqrt(Hz)
- Application
  - Ground application: EM-sounding
  - Space application: sounding rockets and all magnetosphere missions since late 60s



**Figure 1.** Representative electric field spectra for various plasma wave phenomena observed in the Earth's magnetosphere.

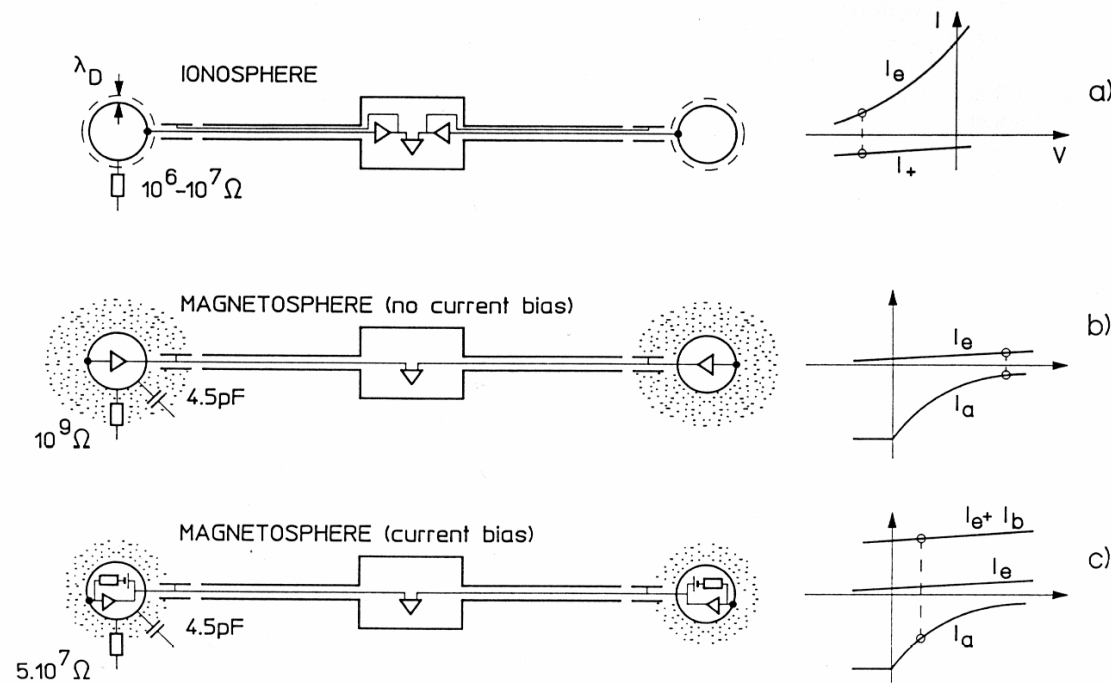
[D.A.Gurnett, 1998]

# Double Probe

- Problem areas

- Large probe separation necessary to increase signal, wire booms are used normal to spin axis, boom for spin axis component difficult

$$I_{plasma-e} + I_{ph-e-ret} + I_{ph-e-sc} + I_{measur} = I_i + I_{ph-e}$$



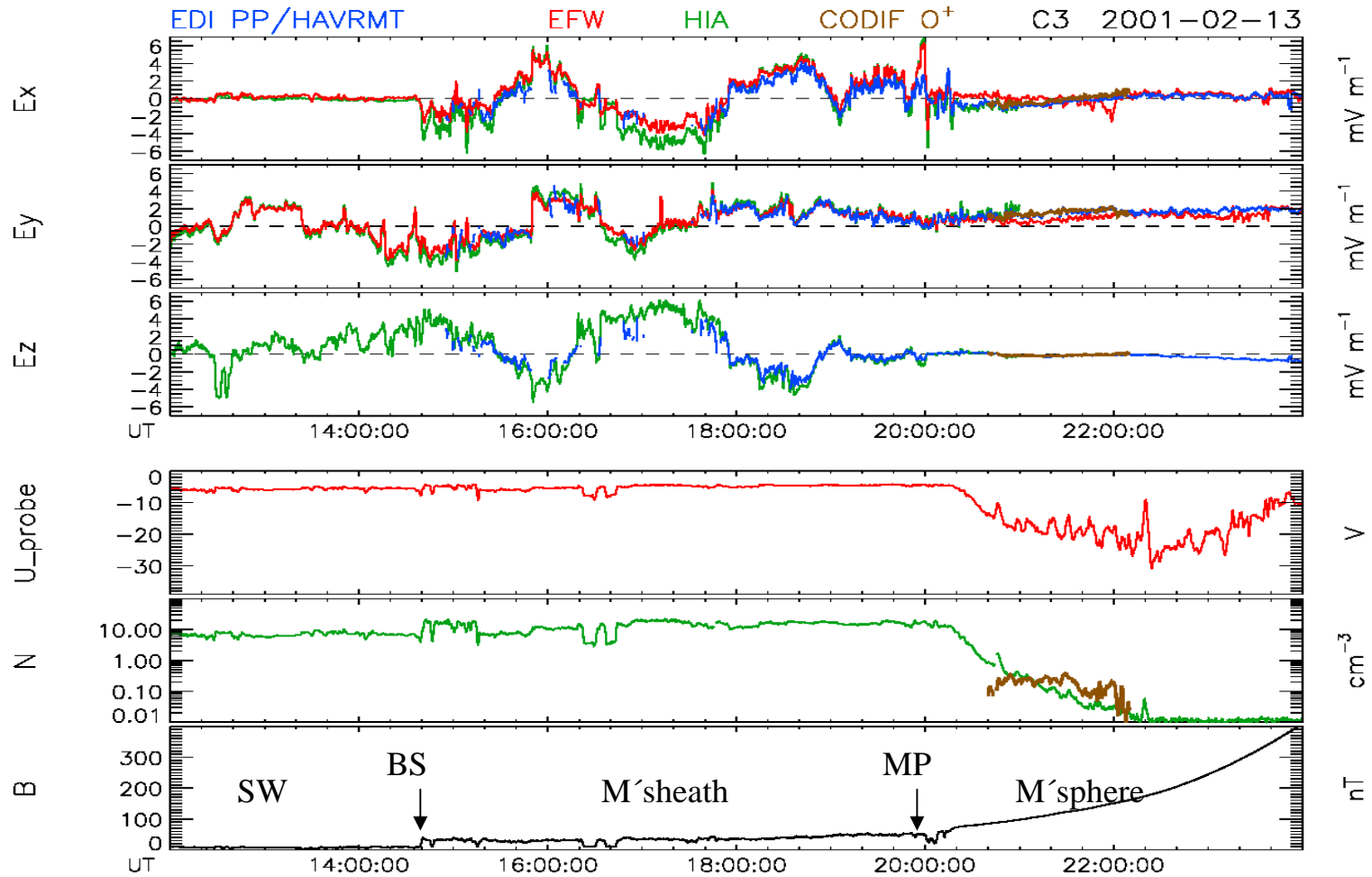
E-field measurement - Double Probe (3)

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[Pedersen et al., 1998]

# *E-Field Examples*



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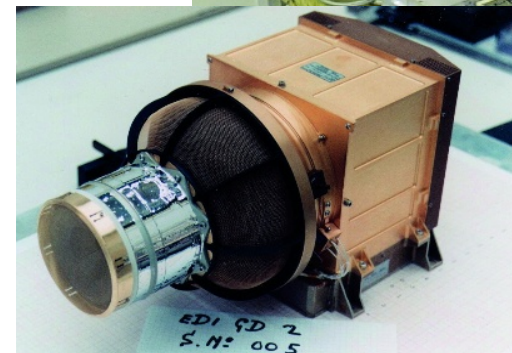
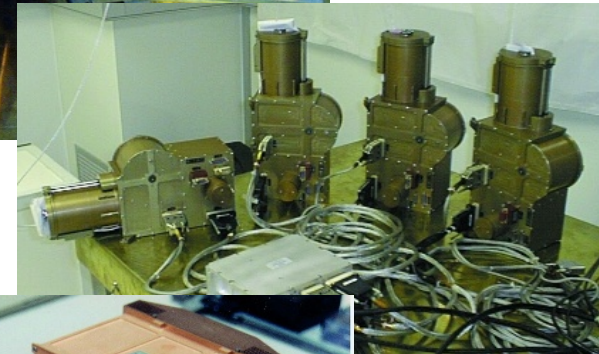
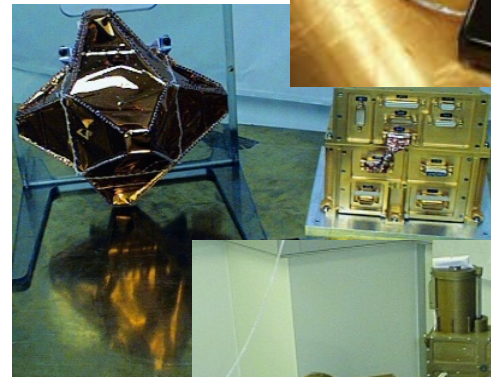
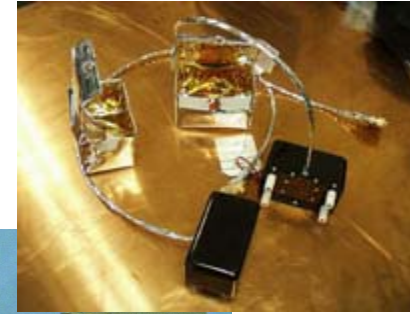
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Examples - E-field Electric Fields from EFW, EDI & CIS; [Eriksson et al., 2003]  
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# *E&M Instrumentation on Cluster*

- Fluxgate Magnetometer, FGM
  - PI: Andre Balogh IC London
  - Dual Sensor (3.1m & 5.1m distance)  
analogue FG
- Search Coil Magnetometer, STAFF
  - PI: Nicole Cornilleau-Wehrin,
  - Three component boom mounted  
sensor
- Double Probe, EFW
  - PI: Mats Andre, IRFU Uppsala
  - Two sets of wire booms in spin plane
- Electron Drift Experiment, EDI
  - PI: Götz Paschmann, MPE Garching
  - Two gun/sensor system

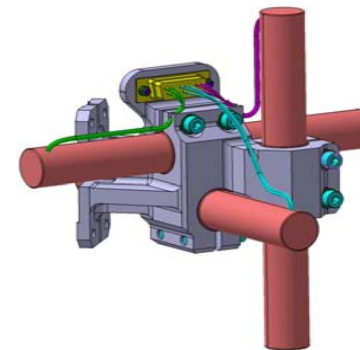


Examples - Cluster  
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# *E&M Instrumentation on Themis*

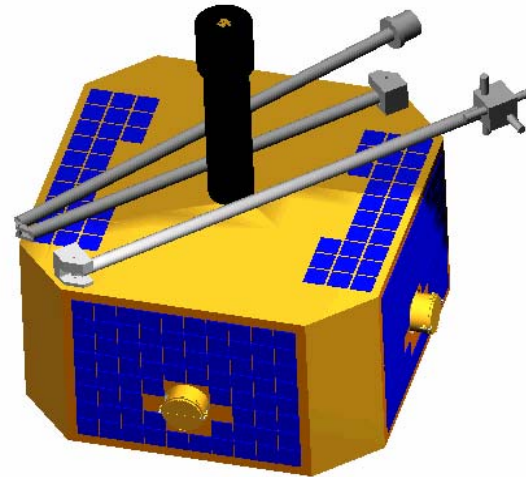
- Fluxgate Magnetometer, FGM
  - K.H. Glassmeier, TU-Braunschweig
  - Single sensor (2m boom), digital electronics
- Search Coil Magnetometer, SCM
  - A.Roux, CEPT
  - The SCM 3-axis antennas are located at the end of 1 meter SCM boom
- Double Probe, EFI
  - J.Bonell, UCB
  - Two sets of wire booms in spin plane
  - One set of axial booms



Examples - Themis  
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# *E&M Sensor Accommodation*

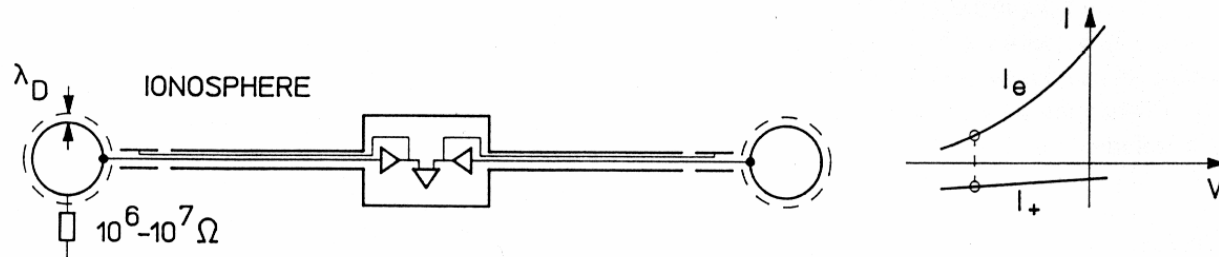
- Boom Geometry
  - 1-3 segment boom (Cluster Double Star, Themis ...)
  - Boom made by spring elements (MMO)
  - Wire Booms (Cluster, Themis ...)
- Release & launch lock mechanism
  - Doors:
    - Pyro (Cluster)
    - Shaped Memory Alloy (Themis)
  - Wires: motor driven
  - Axials: Stacers
- Deployment force
  - Centrifugal force
  - Motor driven



# Double Probe Theory

- Debye shielding,  $\lambda_D = (kT_e / Ne^2)^{1/2}$ 
  - Homework#1: Derive potential at 1-D grid is  $\phi = \phi_0 \exp(-|x| / \lambda_D)$
- Ambient electron current collected by probe
  - $I_e = I_{e0} \exp(V/V_e)$ ,  $V < 0$  where:  $V_e = kT_e / e$
  - $I_e = I_{e0} (1 + V/V_e)$ ,  $V > 0$  where:  $I_{e0} = -A e n_e \sqrt{kT_e / 2\pi m_e}$
  - Homework #2: Prove this
- Ambient ion current collected by probe
  - $I_i = I_{i0} (1 - V/V_e)$ ,  $V < 0$
  - $I_i = I_{i0} \exp(-V/V_e)$ ,  $V > 0$  where:  $I_{i0} = I_{e0} / \sqrt{m_i / m_e}$
  - Homework #3: assuming  $I_i + I_e = 0$ , show that
    - Floating potential =  $kT_e \sqrt{m_i / m_e}$
    - Ion saturation current, when  $V \ll 0$ , is:  $I_{sat} = I_{i0} = A e n_e \sqrt{kT_e / 2\pi m_i}$
- Once  $I_{sat}$  is found, slope gives  $kT_e$ , and A gives  $n_e$ 
  - This is operational principle of Langmuir probes

# Ionospheric measurements



- $\lambda_D \sim 1\text{cm}$ ;  $R = (\delta I / \delta V)^{-1} \sim 10^7 \Omega$ ;  $C \sim 10\text{pF}$  (sphere + stray)
- Probe charges negative due to  $V_e > V_i$
- Voltage set by:
  - Balance of ambient electron ( $I_e$ ) and ion ( $I_+$ ) currents
  - Note: other currents are smaller:  $I_{\text{plasma-e}} + \cancel{I_{\text{ph-e-ret}}} + \cancel{I_{\text{ph-e-sc}}} + \cancel{I_{\text{measur}}} = I_i + \cancel{I_{\text{ph-e}}}$
- High impedance ( $>10\text{M}\Omega$ ) measurement easy
- Large fields  $\Rightarrow$  Short booms  $\Rightarrow$  Preamp on board
- Cylindrical are as just good as spherical sensors

# Ionospheric measurements: problems

## **1<sup>st</sup> order problem**

- Velocity wake

## **Mitigation**

Avoid: 3axis stabilized spacecraft

## **2<sup>nd</sup> order problems**

- Photoemission
- Body photoelectrons

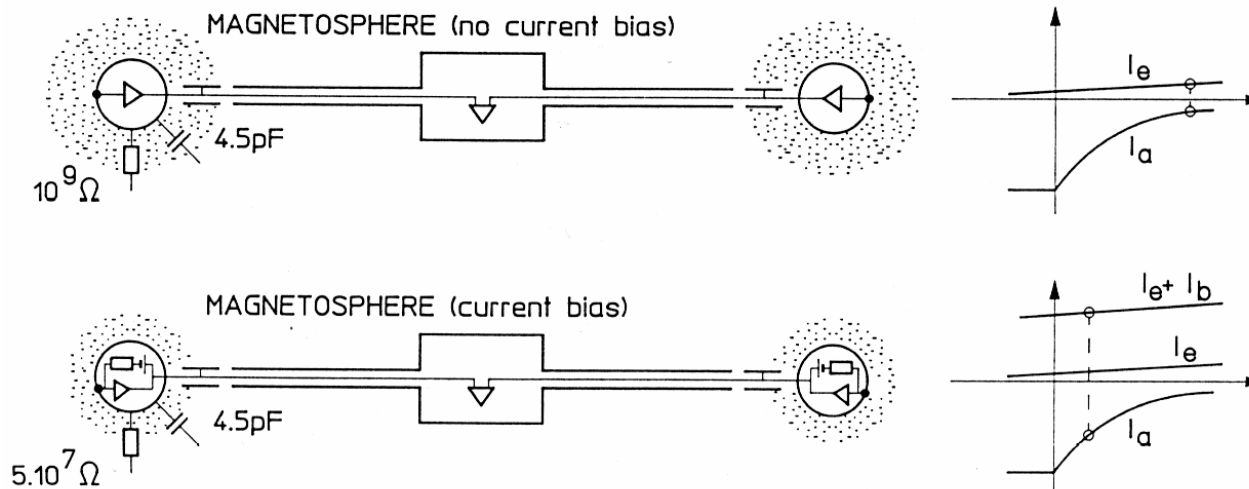
## **Mitigation**

Use spherical sensors

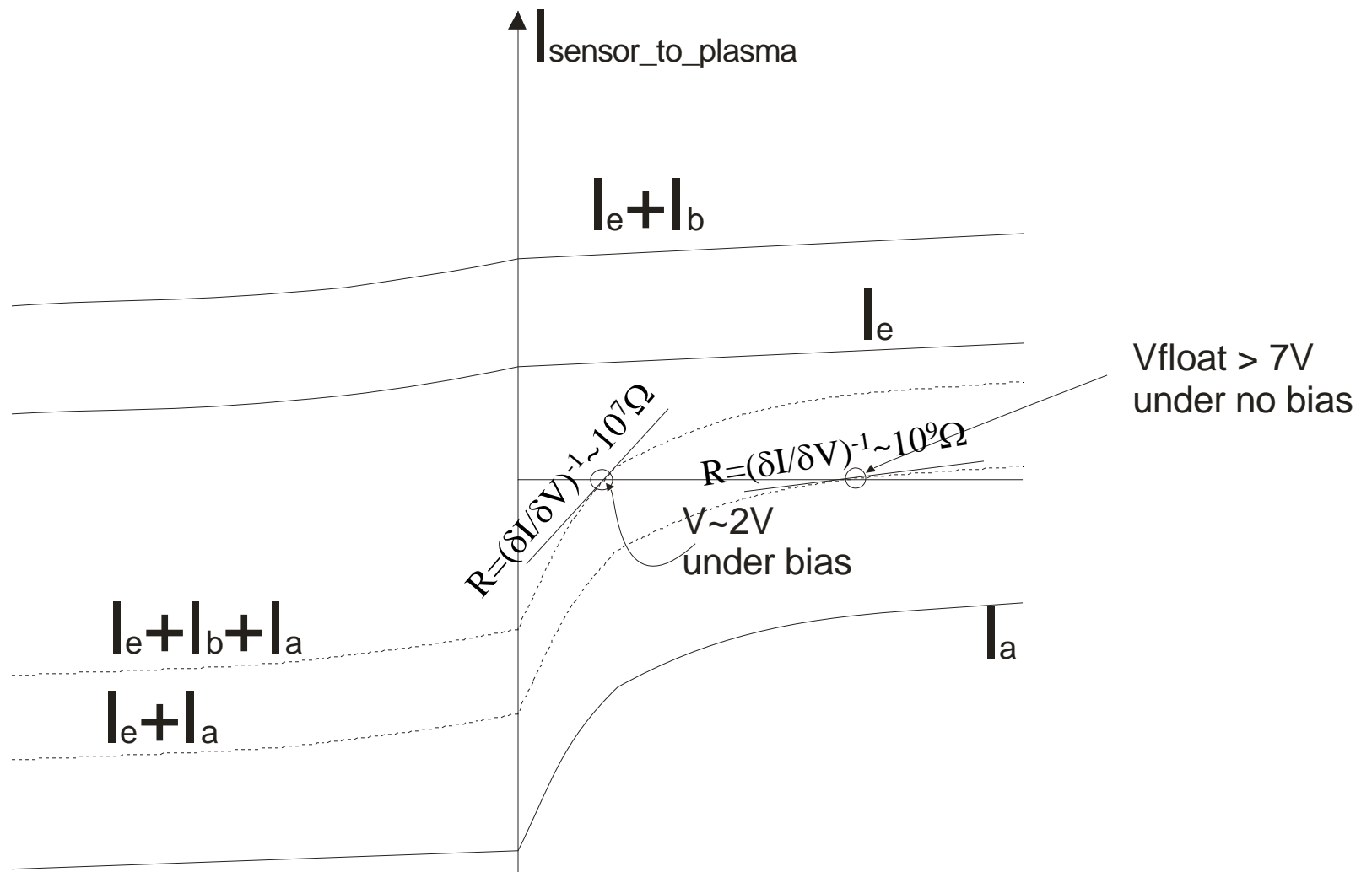
Repel with guards

Others (see Maynard, N. Table 1).

# Magnetospheric measurements



- $\lambda_D \sim 500 \text{ m}$ ;  $R = (\delta I / \delta V)^{-1} \sim 10^9 \Omega$ ;  $C \sim 4.5 \text{ pF}$  (sphere)
- Probe positive due to photoemission ( $I_a$ ).  $I_e$  is from plasma.
  - Other currents are (or made) negligible:  $I_{\text{plasma-e}} + I_{\text{ph-e-ret}} + I_{\text{ph-e-sc}} + I_{\text{measur}} = I_i + I_{\text{ph-e}}$
  - In practice 1<sup>st</sup> order error (and correction) comes from  $I_{\text{ph-e-sc}}$  and  $I_{\text{ph-e-ret}}$
- Resistive coupling to plasma through photoemission
  - Bias current
    - reduces dynamic resistance
    - reduces potential w/r/t plasma
    - makes measurement feasible unless bias





# Magnetospheric measurements

- Photoelectron current: analytical
  - $I_a = I_{a0}, V < 0, I_{a0} = (S/4) j_{a0}$ ; where:
    - $S$  = sphere surface area
    - $j_{a0}$  depends on materials and ambient plasma
  - $I_a = I_{a0}(1 + V/V_{ph})\exp(-V/V_e), V_{ph} = kT_{ph}/e, V > 0$
  - Further reading: Grard, 1973
- Photoelectron current: empirical (x4)
  - $j_a = 80(\mu\text{Am}^{-2})\exp(-V/2) + 3(\mu\text{Am}^{-2})\exp(-V/7.5)$ ;  $V$  in Volts
  - Further reading: Laakso and Pedersen, 1994
- Objective:
  - Bias current such that potential only 1-2Volts above plasma
- Coupling:
  - Resistive below  $1/R_s C_s \sim 100\text{Hz}$ ; capacitive above it

# Magnetospheric measurements: problems

## 1<sup>st</sup> order problems

- Sensor asymmetric photoemission
- Body photoelectrons
- Ion wake
- Axial: no DC

## Mitigation

Constrain s/c pot with guards  
Repel with guards  
Longer booms  
Make symmetric spacecraft  
Electrostatically clean  
Obtain  $E_{\text{axial}}$  from  $\mathbf{E} \cdot \mathbf{B} = 0$

## 2<sup>nd</sup> order problems

- Shielding of external field by sc
- Magnetic wake

## Mitigation

Use negative guards, or  
...live with it  
Live with it